

Internal Loads

Before you can design a structural member, it is necessary to understand the forces and moments that act on it.

Force
acting on the beam

Compression
Pushes the material together

Tension
Stretches the material

Internal Loads

Axial Force - Buckling

Internal Loads

Axial Buckling Force

Internal Loads

Shear Force

Internal Loads

Horizontal Shear Force

Internal Loads

Compression and Tension Forces
Bending Moment

Internal Loads

Internal Loadings Developed in Structural Members

- Before a structural member can be sized or designed, the forces and moments that act on the member must be determined.
- In this section, we will develop methods to calculate the forces and moment at any point along the member's axis.

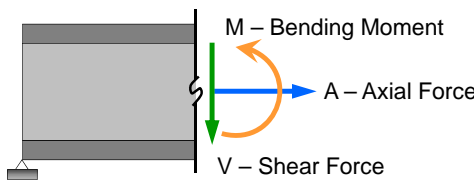
Internal Loads

Internal Loadings at a Specified Point

- In general, the internal load at a specified point can be determined by the **method of sections**
- Typically, coplanar structures are subject to internal loadings that may consist of an axial force **A**, a shear force **V**, and the bending moment **M**

Internal Loads

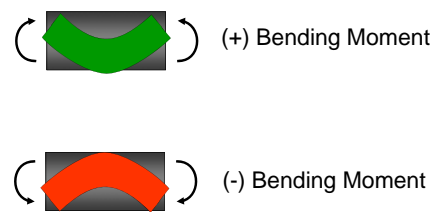
Internal Loadings at a Specified Point



The diagram shows a vertical structural member with a pin support at the bottom. At a specific point, three internal forces are shown: a blue arrow pointing right labeled 'A - Axial Force', a green arrow pointing down labeled 'V - Shear Force', and an orange curved arrow labeled 'M - Bending Moment'.

Internal Loads

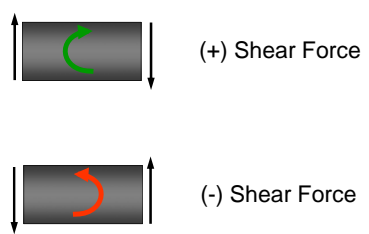
Internal Loadings at a Specified Point



Two diagrams illustrate bending moments. The top diagram shows a green curved arrow pointing upwards, labeled '(+) Bending Moment'. The bottom diagram shows a red curved arrow pointing downwards, labeled '(-) Bending Moment'.

Internal Loads

Internal Loadings at a Specified Point

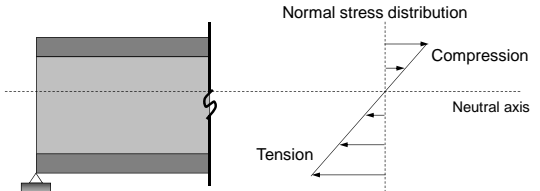


Two diagrams illustrate shear forces. The top diagram shows a green curved arrow pointing to the right, labeled '(+) Shear Force'. The bottom diagram shows a red curved arrow pointing to the left, labeled '(-) Shear Force'.

Internal Loads

Internal Loadings at a Specified Point

These three types of loadings are used to represent the **stress distribution** acting over the member cross-section



The diagram shows a structural member with a pin support at the bottom. To its right, a cross-section is shown with a normal stress distribution. A horizontal dashed line represents the 'Neutral axis'. Above the neutral axis, the stress is labeled 'Compression' with arrows pointing towards the axis. Below the neutral axis, the stress is labeled 'Tension' with arrows pointing away from the axis.

Internal Loads

Procedure for analysis - the following is a procedure for determining the internal forces in a member using the method of sections:

1. Before the member is "cut" or sectioned, determine the support reactions for the structure.
2. Keeping all external loadings in their exact locations, make an imaginary "cut" through the member at the point where the internal loading is desired.

Draw the corresponding free-body diagram of one of the "cut" segments indicating the unknown reactions **A**, **V**, and **M** acting in their positive (+) directions

Internal Loads

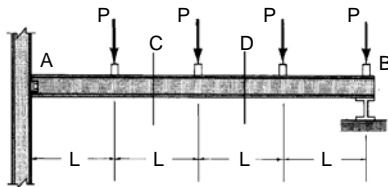
Procedure for analysis - the following is a procedure for determining the internal forces in a member using the method of sections:

3. Apply the three equations of equilibrium.

In most cases, the moment equation should be summed at the cut section about the *centroid* of the member cross-section to eliminate the unknowns **A** and **V**.

Internal Loads

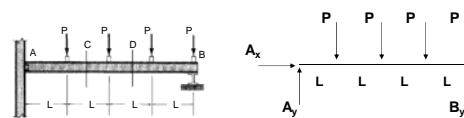
Example: Consider the following beam



Determine the shear and moment in the floor girder at points C and D (both points are at the center of each span L).

Internal Loads

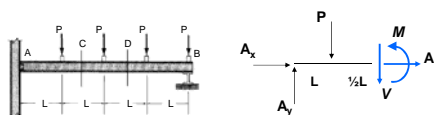
Example: Consider the following beam



$$\begin{aligned} \sum^+ M_A = 0 &= -P(L + 2L + 3L + 4L) + B_y(4L) & B_y &= 2.5P \\ \sum^+ F_y = 0 &= A_y + B_y - 4P & A_y &= 1.5P \\ \sum^+ F_x = 0 &= A_x & A_x &= 0 \end{aligned}$$

Internal Loads

Cut the beam at point C



$$\begin{aligned} \sum^+ M_{cut} = 0 &= M + P\frac{L}{2} - A_y\frac{3L}{2} & M &= 1.75PL \\ \sum^+ F_y = 0 &= A_y - V - P & V &= 0.5P \\ \sum^+ F_x = 0 &= A & A &= 0 \end{aligned}$$

Internal Loads

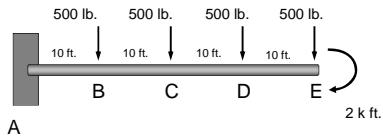
Cut the beam at point D



$$\begin{aligned} \sum^+ M_{cut} = 0 &= M + P\frac{L}{2} + P\frac{3L}{2} - A_y\frac{5L}{2} & M &= 1.75PL \\ \sum^+ F_y = 0 &= A_y - V - P - P & V &= -0.5P \\ \sum^+ F_x = 0 &= A & A &= 0 \end{aligned}$$

Internal Loads

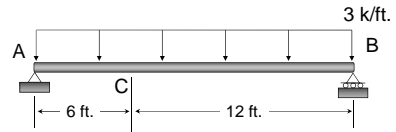
Example: Consider the following beam



Determine the internal shear and moment in the cantilever beam shown above at a section passing through point C.

Internal Loads

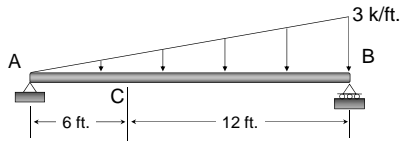
Example: Consider the following beam



Determine the internal shear and moment at a section passing through point C.

Internal Loads

Example: Consider the following beam



Determine the internal shear and moment in the at a section passing through point C.

End of Internal Loads – Part 1

Any questions?

